



The Standard
AIRCRAFT LOG

ASA-SA-2

Aircraft Record General Information

Manufacturer

Piper

Model

PA28-236

Serial

28-801145

Registration Number

N9295M

Date of Manufacture

1980

Engine(s) currently installed:

Manufacturer

Lycoming

Model

O540-J3ASD

Serial

L22502-70A

Manufacturer

Model

Serial

Propeller(s) currently installed:

Manufacturer

Hartzell

Model

HC-F27R-1

HUB Model

E8468A-4R

Serial

C1M 608

Serial

Blade Model

Serial

E61053

Serial

Serial

Blade Model

Serial

E60645

Serial

Serial

YEAR:	RECORDING TACH TIME	TODAYS FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations
DATE				<p>Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)</p>

WAB

N929SM DATE: 12-13-02 MODEL: PA-28-236 S/N: 28-80145
 HOBBS: 860.43 TT: 860.43 WORKORDER: 3283

Altimeter Transponder Test Due Now. VFR ONLY. Change N# CX-BMZ to N929SM.
 ELT Battery due 10-04. Checked pulleys, cables, brackets, hinge points, & lube. Check lights.
 serviced battery, brake reservoir, packed wheel bearings, & cleaned fuel screens. Repaired wheel fairings.
 Replaced rudder with p/n 65342-002. s/n 771614. Repaired by Williams Airmotive, Inc. See Yellow Tag.
 Checked balance after painted. Installed STC SA01071LA by replacing tail cone fairing assembly. See
 337 form. Installed STC SA71GL Brackett Air Filter. See 337 form. Installed wing tips & high intensity
 discharge lights. See 337 form. Sand, prime, paint fuselage using JET GLO CM0570535 Matherhorne white,
 U00550 deep red, U00293 jade mist green, & U00243 buckskin. Aircraft Weighed. See Weight & Balance for
 details. Replaced fire extinguisher. Sealed windshield & windows. Replaced vinyl on glare shield. Replaced
 vacuum regulator filter. Sealed left & right main struts & serviced. Replaced axle in nose wheel, resealed left
 brake cylinder & replaced right brake assembly p/n 30-55, replaced brake hose & serviced brakes. Replaced
 fuel cap gaskets p/n 462-056. Replaced right & left fuel drains. Replaced EGT Probe p/n 86255 & EGT Gauge
 p/n 46150. Complied with SB977 by checking rear wing attach fittings condition. Complied with SB1006 on
 fuel tanks by replacing fuel probe gaskets. Complied with the following AD's: 93-05-06 on gerdes ignition
 switch by installing Kit A3600, 95-26-13 oil cooler hoses due 10-2005. See AD folder for complete list.
 Cleaned inside & out. Ground run opps check good no leaks noted. I certify this Aircraft has been Inspected.
 In Accordance with a Annual/100 Hour Inspection and was determined to be in Airworthy Condition.

William D. Burchfiel

William D. Burchfiel
 AP4148815531A
 Aircraft Technicians, Inc.

YEAR: <u>2002</u> DATE	RECORDING TACH TIME	TODAYS FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
<u>12/14/02</u>				I inspected this Aircraft and have issued a <u>STANDARD</u> Airworthiness Certificate. Dated <u>12/14/02</u> Signed <u>Donald W. Poore</u> <u>DART-200077-80</u>

YEAR:

RECORDING
TACH
TIMETODAYS
FLIGHTTOTAL
TIME IN
SERVICE**Description of Inspections, Tests, Repairs and Alterations**

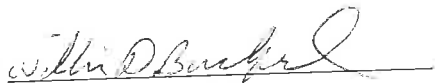
Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)

DATE

N929SM DATE: 12-16-02 MODEL: PA-28-236 S/N: 28-801145
 HOBBS: 860.43 TT: 860.43 WORKORDER: 3283
 Installed new nose gear stop assembly p/n 87469-02 & steering rod assembly
 P/n 87468-020 per Piper Service Kit 764-975V. Weight & Balance Revised.

Gross Weight	3000.00
Empty Weight	1776.00
New C.G.:	83.30
New Moment:	147943.56
New Useful Load:	1224.00

Signed



William D. Burchfiel

AP414881553IA

Aircraft Technicians, Inc.

Dated:

12/16/02

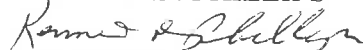
YEAR: DATE	RECORDING TACH TIME	TODAYS FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
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N929SM DATE 1 APR 04
 MODEL:PA28-236 SN 28-8011145
 TACH: 891.50
 TT: 891.50

ALTIMETER / TRANSPONDER TEST DUE NOW. VFR ONLY
 ELT BATTERY : DUE OCT-04. AIRCRAFT INSPECTION AND LUBE
 ACCORDING TO PIPER SERVICE MANUEL.AND FAR43.13 APX.D.

I CERTIFY THAT THIS A/C HAS BEEN INSPECTED IN ACCORDANCE
 WITH AN ANNUAL INSPECTION AS PER PIPER MANUAL AND
 FAR43.13 APX.D. AND WAS DETERMINED TO BE IN AN AIRWORTHY
 CONDITION

KENNETH D. PHILLIPS



AP412800134IA

YEAR: DATE	RECORDING TACH TIME	TODAYS FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
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1 Sep 05 924.9 924.9

N929SM 1 SEPTEMBER 05
 TACH 924.9
 TTAC: 924.9 PERFORMED ANNUAL INSPECTION

ALTIMETER AND TRANSPONDER TEST DUE NOW, VFR ONLY.
 GENERAL HULL INSPECTION CHECKED PULLIES, CABLES
 BELL CRANKS, HINGE POINTS, LUBED A/C, CHECKED WHEEL BEARINGS
 AND BRAKES. INSPECTED ELT IN ACCORDANCE WITH FAR 91.207 (a)
 (1) (d) REPLACED BATTERY DUE. OCT 07. I CERTIFY THAT THIS
 AIRCRAFT HAS BEEN INSPECTED IN ACCORDANCE WITH AN ANNUAL
 INSPECTION AS PER FAR 43.13 APX D AND WAS DETERMINED TO BE IN
 AN AIRWORTHY CONDITION.

SEE AD PROFILE FOR AD'S.

KENNETH PHILLIPS
Kenneth Phillips
 AP4128001341A

ELT BATTERY DUE OCT / 07

TRANSPONDER AND STATIC SYSTEM TEST DUE NOW. FAR 43 APX E & F

TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
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924.9

N929SM 1 SEPTEMBER 05

TACH 924.9

TTAC: 924.9 PERFORMED ANNUAL INSPECTION

ALTIMETER AND TRANSponder TEST DUE NOW, VFR ONLY GENERAL HULL INSPECTION CHECKED THRU LINES, CABLES, WELLS, CRANKS, HINGE POINTS, LUBED A/C, CHECKED WHEEL BEARINGS AND BRAKES. INSPECTED ELT IN ACCORDANCE WITH FAR 91.209 (b) (1) (2) REPLACED BATTERY DUE OCT 07. I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED IN ACCORDANCE WITH AN ANNUAL INSPECTION AS PER FAR 43.15 APX D AND WAS DETERMINED TO BE IN AN AIRWORTHY CONDITION.

SEE AD PROFILE FOR AD'S.

ELT BATTERY DUE: OCT / 07

TRANSponder AND STATIC SYSTEM TEST DUE NOW, FAR 43 APX E & F.

KENNETH PHILLIS
Kenneth Phillis
 AT 4125161341A

YEAR:	RECORDING TACH TIME	TODAYS FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
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2006
DATE

2-4-06

July 27, 2006
N-929SM

Logbook Entry
PA-28-236

Tach: 937.33
S/N 28-8011145

Installed new G-35 Aircraft battery.
Cleaned box & cable ends. Jack shape #P 1912 rto
eps ok good.

A) Removed the following:
 1) Flightcom DVR 300i Recorder S/N 22137
 2) Bendix 102A Audio Amplifier S/N 12649
 3) Bendix/King KT-78A Transponder P/N 066-1062-01 S/N 8579

B) Installed the following:
 1) PS Engineering PMA 8000B Audio Panel/Mkr BC N Rec/ Intercom S/N M02288
 2) Garmin GNS-430 Comm/VOR/LOC/Glideslope/GPS P/N 011-00280-10 S/N 97130301 TSO C34e, C36e, C37d, C40e, C129a-A1
 3) Garmin GA-56 GPS Antenna P/N 011-00134-00 S/N 59340929 TSO C129a- A1
 4) Garmin GI-106A Indicator P/N 013-00049-01 S/N E06-10476 TSO C34e, C36e, C40a
 5) ACK A-30 Blind Encoder S/N 100547 TSO C88a
 6) Comant CI-122 Comm Antenna S/N 72460 TSO C37d, C38d
 7) Comant CI-505 Nav/Glideslope Antenna Splitter S/N 128323 TSO C34e, C36e, C40a
 8) Comant CI-102 Marker Beacon Antenna S/N 111303 TOS C35d
 9) Garmin GTX-327 Transponder P/N 011-00490-00 S/N 83725027 TSO C74e
 10) Davtron M800-14V-B Chronometer S/N 22277

C) The above is installed with reference to the following:
 1) Garmin 400 Series Installation Manual P/N 190-00140-02 Revision Q Dated 10-13-05.
 2) Garmin GI-106A Installation Manual P/N 190-00180-00 Revision C Dated April 2001.
 3) PS Engineering PMA-8000B Installation Manual P/N 200-890-0100 Revision 5 Dated 5-30-06.
 4) ACK A-30 Installation Manual P/N A-30M Revision 3 Dated 7/9-2003.
 5) Garmin GTX-327 Installation Manual P/N 190-00187-02 Revision L Dated June 7, 2006.

CO,VT →

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DATE				
				<p>6) Davtron 800 Series Installation Drawing #800-005 Dated 8-16-86.</p> <p>7) FAA AC43.13-2A Chapter 1 Paragraph 4 Materials, 5 Fabrication, 6 Fasteners, 7 Protection Against Deterioration, 8 Provisions for Inspection, 9 Effects on Weight and Balance, 10 Effects on Safe Operation, 11 Controls and Indicators, 12 Placarding, Chapter 2 Paragraph 21 Installation, 22 Equipment Manufacturer's Instructions, 23 Instrument Panel Mounting, 27 Electrical Systems. Chapter 3 Paragraph 36 Performance, 37 VHF Antenna Whip, 41 Marker Beacon Antenna.</p> <p>8) FAA AC20-138 Section 8c (2) i thru iv.</p> <p>9) FAA AC90-100 which TSO C129a-A1 is in compliance with and meets performance and functional requirements of this AC.</p> <p>D) Description of work performed:</p> <p>1) The existing Comm1/Nav1 (King KX-170B/K1-208) were relocated to the #2 Position.</p> <p>2) The PS Engineering PMA 8000B Audio Panel, Marker Beacon Receiver, Intercom is located at the top of the left avionics stack at F.S. 60.0. The PMA-8000B is connected to the avionics buss thru a 5 amp circuit breaker which is placarded "Audio Panel". The PMA-8000B is connected to the Garmin GNS-430 Mic & Audio Outputs as #1, the existing King KX-170B Nav/Comm Mic & Audio Outputs as #2 and the existing King KR-85 ADF Audio. The PMA-8000B Intercom is installed as a Four-Place (Pilot & Co-Pilot Pass 1 & 2) System with Pilot & Co-Pilot Yoke mounted mic switches. The PMA-8000B Marker Beacon Receiver is connected to the existing CI-102 Marker Beacon Antenna System thru RG-400 Coaxial Cable. The CI-102 is located on the bottom of the fuselage at F.S. 114.0. A 13.5"X10.5"X.063 approximate Backing Plate was fabricated from 2024T-3 Alclad Aluminum for the CI-102. The Backing Plate is secured with 37 ea. MS20470AD4 Rivets. The CI-102 is sealed with MIL-A-46106 Sealant.</p> <p>3) The GNS-430 Com/VOR/G/S/GPS is located in the Left Radio Stack at F.S.58.0 and is installed as the #1 System. The GNS-430 Comm power is connected to the Avionics Buss thru a 10 amp circuit breaker which is placarded "Comm 1" The GNS-430 VOR/G/S/GPS is connected to the Avionics Buss thru a 5 amp circuit breaker which is placarded "GPS".</p>

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DATE				
				<p>The GNS-430 is wired to display Nav or GPS, Deviation, Course, and Glideslope information onto the GI-106A Indicator which is located in the top right corner of the pilots instrument panel at F.S. 60.0. The GNS-430 is connected to the existing Nav/Glideslope Antenna System thru the CI-505 Antenna Coupler.</p> <p>The GNS-430 is connected to the GA-56 GPS Antenna with RG-400 Coaxial Cable. The GA-56 is located on top of the fuselage at F.S. 93.5 utilizing the existing factory antenna doubler. The GNS-430 Comm is connected to the Comant CI-122 Comm Antenna with RG-400 Coaxial Cable. The CI-122 is located on the bottom of the fuselage at F.S. 114.0. A 10.5" X .063 approximate Backing Plate was fabricated from 2024T-3 Alclad Aluminum and is secured with 32 ea MS20470AD4 rivets. The CI-122 and the GA-56 are sealed with MIL-A-46106 Sealant.</p> <p>The GNS-430 Comm Mic/Phone and Nav Audio are connected to the respective #1 Comm/Nav Audio inputs of the Garmin GMA-340 Audio Panel. The Mode-C Altitude Inputs of the GNS-430 are connected to the ACK A30 RS-232 Mode C Output. The GNS-430 is not coupled to any Automatic Flight Control System.</p> <p>The Garmin GTX-327 Transponder is located in the Right Avionics Stack at F.S. 60.0. The GTX-327 is connected to the Avionics Buss thru a 3 Amp Circuit Breaker which is placarded "Transponder".</p> <p>The ACK A-30 Encoder is located behind the Co-Pilots Panel at F.S. 54.0. The Encoder is connected to the Avionics Buss thru a 3 Amp Circuit Breaker which is placarded "Enc".</p> <p>The Garmin GTX-327 is connected to the RS-232 Mode-C Output of the A-30 Encoder Aircraft Ground and Dimmer System. The GTX-327 is connected to the Existing Transponder Antenna System.</p> <p>The A-30 Encoder is connected to Aircraft Ground and to the GNS-430 RS-232 Mode-C Input. The A-30 is also connected to the existing Static System.</p> <p>5) The Davtron 800 Chronometer is located in the upper left corner of the Pilots Panel at F.S. 62.0.</p> <p>The Davtron 800 is connected to the existing Piper Clock D.C. Battery Voltage, Aircraft Ground and Dimmer.</p>

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				<p>E) Operational Checks:</p> <ol style="list-style-type: none"> 1) The equipment installed was Ground and Flight checked per manufacturers check-out procedures and found to function properly. 2) The Altimeter System and Altitude Reporting equipment was tested per FAR 91.411 in accordance with FAR 43 Appendix E Paragraph (a) to 2,000' (b) United 5934PD-1 S/N T1204 to 20,000ft and (c) to 20,000ft. 3) The Transponder System operation was tested and meets requirement of FAR 91.413 as prescribed in Appendix F of Part 43 on the Garmin GTX-327 Transponder above. <p>F) Approval Status:</p> <ol style="list-style-type: none"> 1) The GNS-430 Comm/VOR/LOC/Glideslope are certified for VFR or IFR Airborne Operations. The GPS portion of the GNS-430 is certified for VFR or IFR Enroute/Terminal/Non-Precision Approach when operated in accordance with the following: <ol style="list-style-type: none"> A) FAA Approved Flight Manual Supplement #PA28236 / 28-8011145/430-1 which was provided for this aircraft and this installation. The FAA Approved Flight Manual Supplement is located in the FAA Approved Airplane Flight Manual of the Piper PA28-236 P/N 761-689 Revision 3 Dated 6-1-78 or later appropriate revision. B) Garmin GNS-430 Pilots Guide and Reference P/N 190-001400-00 Revision A Dated October 1998 or later appropriate revision. This manual must be immediately available to the flight crew whenever Navigation is predicated on the use of the GNS-430 System. 2) The above FAA Approved Flight Manual Supplement for the GNS-430 Dated <u>July 27, 2006</u> which was provided, must be maintained in the aircraft at all times while the equipment is in use. The above systems were functional tested and found to be compatible with the above listed Flight Manual Supplement.

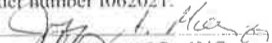
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H) General

- 1) Weight and Balance computed, Equipment list revised. See 337 form dated same.
- 2) Compass Swung.
- 3) Instructions for Continued Airworthiness attached to the 337 form.
The ICA is now a part of the Aircraft Inspection/Maintenance Requirements.
- 4) Details are on file at this Repair Station under work order number 1062021.

End


STAR AVIONICS, INC.
 LUTELL FIELD
 CHATTANOOGA, TN 37421
 FAA LICENSED REPAIR STATION # TA 48517M

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DATE				

N929SM

DATE: 25 SEP 06

100 HR INSPECTION

TACH: 944.0

TT: 944.0

100 HR INSPECTION PERFORMED IAW PIPER SERVICE MANUAL. JACKED A/C CHEK WHEELS AND BRAKES . CHECKED AND SERVICED BATTERY LUBED A/C . . ELT CRASH TEST PERFORMED IAW FAR 91.207 CHECKS GOOD ELT BATTERY DUE OCT 07. INSTALLED NEW TACH CABLE PIPER P/N 486-594

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED IAW AN ANNUAL INSPECTION AS PER FAR 43.13 APX D AND WAS DETERMINED TO BE IN AN AIRWORTHY CONDITION. SEE AD PROFILE FOR LIST OF AD COMPLIANCE

KENNETH D PHILLIPS

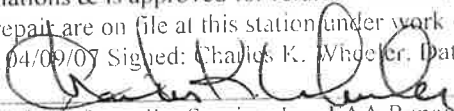
Kenneth D Phillips
AP412800134IA

4 Apr 07	Removed	Propeller for over haul - sent to ATL Propeller		
Tach.	1011.0			<i>Kenneth D Phillips</i> AP412800134

YEAR: DATE	RECORDING TACH TIME	TOTAL FLIGHT	TIME IN SERVICE	Entries made at Technician or Repair Facility. (See back)
4/20/07				VOR Check AT VXV VOR @ 25 HME in 074° RADIAL ERROR - 2°

TACH - 1011.0

Make: Hartzell Model # HC-F2YR-1F/F8468A-4R S/N CM608.
 The propeller identified above was repaired & inspected I/A/W
 current maintenance rules of the Federal Aviation Administration
 regulations & is approved for return to service. Pertinent details of
 the repair are on file at this station under work order # 041007.
 date 04/09/07 Signed: Charles K. Wheeler, Date of 04/11/07.



For Atlanta Propeller Service, Inc. FAA Repair Station #
 DR4R138M. 4787 Clark Howell Hwy, College Park, GA 30349

Carl Nunn

TACH: 1011.0
 15A104 - Installed off Prop
 from A7L-Propeller, Torque
 Bolts & Safeties with HOS of rig.
 Aircraft Returns to Service

James Kelly
 A41255139

YEAR:	RECORDING TACH TIME	TODAYS FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alteration Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
DATE				

N929SM

DATE: 18 SEP 07

TACH: 1033.40

A/C TT: 1033.40

100 HR INSPECTION IAW PIPER MAINTENANCE. LUBE A/C . . AD COMPLIANCE WITH T-DATA 2007-18 TRANSPONDER AND PITOT STATIC. DUE JUL 09 . INSTALLED PROP GUARD ON PROPELLER IAW STCSP00582NY *Air Filter BA 2905 changed (✓)*

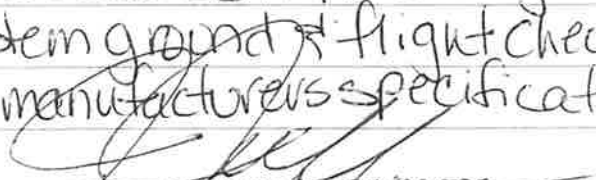
ELT INSPECTION IAW FAR91.207(D) CHECKS GOOD. ELT BATTERY DUE SEP 09

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED IN ACCORDANCE WITH AN ANNUAL INSPECTION AS PER FAR 43.13 APX D AND WAS DETERMINED TO BE IN AN AIRWORTHY CONDITION..
SEE AD PROFILE FOR LIST OF Ads

KENNETH D PHILLIPS

Kenneth D Phillips
AP4128001341A

YEAR: DATE	RECORDING TACH TIME	TODAYS FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
9/20/07	TACH: 1035.94.			Sent GNS430 bin 97130361 to Garmin for upgrade to WAAS - reinstalled unit. Replaced existing GA56 antenna with GA35 WAAS/GPS antenna. For details on installation see 337. No change in weight & balance. System ground & flight checks ok in accordance with manufacturers specifications.


AIR ASSOCIATES INC
CRS# LS6R1101

DATE: 7 DEC 07
TACH: 1047.21

DG REMOVED FOR O/H AND SENT TO RUDY A/C INSTRUMENTS, RUDY, AK.
SIGMA TEK DG P/N 4000B S/N T43753C INSTALLED IN AIRCRAFT 1AW PIPER SERVICE
MANUAL AND RUDY INSTRUCTIONS , GROUND OPS CHECK GOOD AIRCRAFT
RETURNED TO SERVICE.

KENNETH D PHILLIPS

AP412800134

**ATE
TAG**

3. Form Tracking Number:

35810

5. Work Order/Contract/Invoice
Number:

35810

1	8. Part Number:	9. Engineering:	10. Qty:	11. Serial/Batch Number:	12. Status / Work:
1	SIGMA TEK DG 4000B		1	T43753C	Overhauled

13. Remarks:

14. Certifies the items above were manufactured in conformity to:

Approved design data and are in condition for safe operation.

Non-approved design data specified in Block 13

19. 14 CFR 43.9 Return to Service Other regulation specified in Block 13

Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished with Title 14, Code of Federal Regulation, part 43 and in respect to that work, the items are approved for return to service.

15. Authorized Signature:

16. Approval/Authorization No.:

20. Authorized Signature:

21. Approval/Authorization No.:

RIZR316K

17. Name (Typed or Printed):

18. Date (m/d/y):

22. Name (Typed or Printed):

23. Date (m/d/y):

Michael McIntosh

Dec 3 2007

User / Installer Responsibilities:

It is important that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block-1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/component/assemblies from the airworthiness authority of the country specified in Block-1.

Statements in Blocks-14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.